

5- Applications on Second and Higher Order Linear ODE

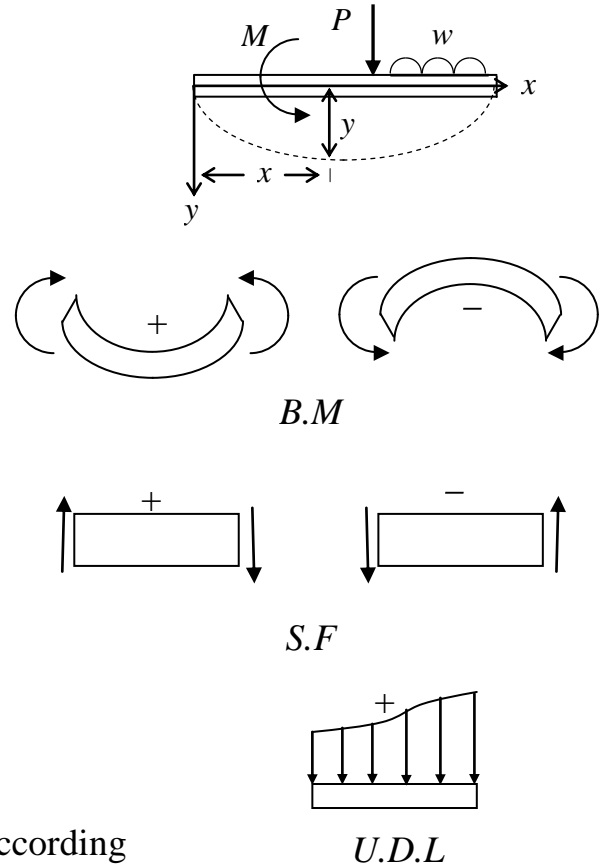
1. Deflection of beams

Differential equations of deflected shapes:

$$EI \cdot \frac{d^2 y}{dx^2} = -M_x,$$

$$\therefore V = \frac{dM}{dx} \Rightarrow \therefore EI \cdot \frac{d^3 y}{dx^3} = -V_x,$$

$$\therefore w = -\frac{dV}{dx} \Rightarrow \therefore EI \cdot \frac{d^4 y}{dx^4} = w_x,$$



where;

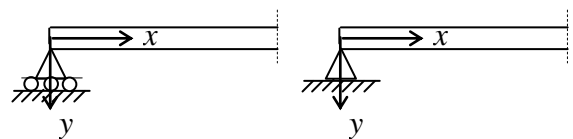
y is the deflection, M_x , V_x , and w_x are the bending moment (B.M), shear force (S.F), and uniformly distributed load (U.D.L) at a section at distance x , respectively, and EI is the rigidity of the cross section of the beam.

The following boundary conditions are stated according to the supporting type,

* Hinge or pin or roller,

No deflection $\Rightarrow y(0) = 0.$

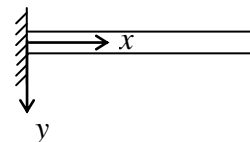
No bending moment $\Rightarrow y''(0) = 0.$



* Fixed or clamped,

No deflection $\Rightarrow y(0) = 0.$

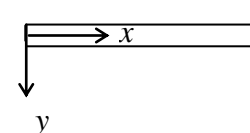
No rotation $\Rightarrow y'(0) = 0.$



* Free end,

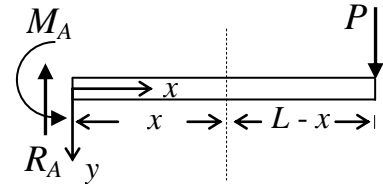
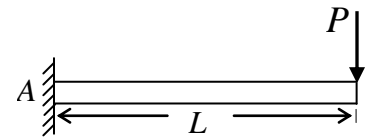
No bending moment $\Rightarrow y''(0) = 0.$

No shear force $\Rightarrow y'''(0) = 0.$



Example 1: A cantilever beam of length L is subjected to a concentrated load P at the free end. Derive and solve the differential equation of the deflection curve of the beam and also find the maximum deflection.

Solution:



$$EI \cdot \frac{d^2 y}{dx^2} = -M_x.$$

$$M_x = -P(L - x),$$

$$\therefore EI \cdot \frac{d^2 y}{dx^2} = -[-P(L - x)],$$

or
$$\frac{d^2 y}{dx^2} = \frac{P}{EI}(L - x).$$

Method I,

Since the right side terms are functions of x only, we can solve the DE by integrating both sides directly,

$$\frac{dy}{dx} = \frac{P}{EI} \left(Lx - \frac{x^2}{2} \right) + C_1,$$

$$y = \frac{P}{EI} \left(\frac{Lx^2}{2} - \frac{x^3}{6} \right) + C_1 x + C_2. \quad \text{(G.S)}$$

Method II,

We can solve the DE as a non-homogeneous linear DE,

$$D^2 y = 0 \quad \Rightarrow \quad m^2 = 0 \quad \Rightarrow \quad m_{1,2} = 0,$$

$$\therefore y_c = C_1 x + C_2.$$

Let $y_p = (A_0 + A_1 x)x^2 \quad \Rightarrow \quad y_p = A_0 x^2 + A_1 x^3,$

$$y'_p = 2A_0 x + 3A_1 x^2 \quad \Rightarrow \quad y''_p = 2A_0 + 6A_1 x,$$

Substituting,

$$2A_0 + 6A_1 x = \frac{P}{EI}(L - x) \quad \Rightarrow \quad 2A_0 + 6A_1 x = \frac{PL}{EI} - \frac{Px}{EI},$$

To determine M_x :

Either from left;

$$M_x = R_A \cdot x - M_A.$$

$$\sum F_y = 0,$$

$$P - R_A = 0 \quad \Rightarrow \quad R_A = P.$$

$$\sum (M)_A = 0,$$

$$M_A - P \cdot L = 0 \quad \Rightarrow \quad M_A = PL.$$

$$\therefore M_x = P \cdot x - PL$$

$$= -P(L - x).$$

Or from right;

$$M_x = -P(L - x).$$

$$\therefore 2A_o = \frac{PL}{EI} \Rightarrow A_o = \frac{PL}{2EI},$$

$$6A_1 = -\frac{P}{EI} \Rightarrow A_1 = -\frac{P}{6EI},$$

$$\therefore y_p = \frac{PL}{2EI}x^2 - \frac{P}{6EI}x^3 = \frac{P}{EI}\left(\frac{Lx^2}{2} - \frac{x^3}{6}\right).$$

$$y = y_c + y_p \Rightarrow y = C_1x + C_2 + \frac{P}{EI}\left(\frac{Lx^2}{2} - \frac{x^3}{6}\right). \quad (\text{G.S})$$

Boundary conditions,

$$1- y(0) = 0 \Rightarrow 0 = 0 + C_2 + 0 \Rightarrow C_2 = 0.$$

$$2- y'(0) = 0, \quad y' = C_1 + \frac{P}{EI}\left(Lx - \frac{x^2}{2}\right) \Rightarrow 0 = C_1 + 0 \Rightarrow C_1 = 0.$$

$$\therefore y = \frac{P}{EI}\left(\frac{Lx^2}{2} - \frac{x^3}{6}\right) \quad \text{or} \quad y = \frac{Px^2}{6EI}(3L - x). \quad (\text{P.S})$$

Maximum deflection of cantilever beams occurs at the free end, that is at $x = L$,

$$y_{\max} = \frac{P(L)^2}{6EI}(3L - L) = \frac{PL^3}{3EI}.$$

Example 2: Find the deflection curve and the reactions for the beam shown below.

Solution:

$$EI \cdot \frac{d^4 y}{dx^4} = w_x.$$

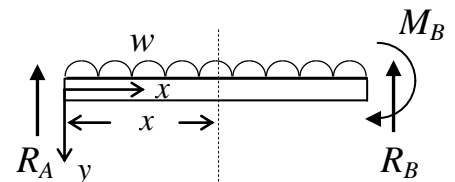
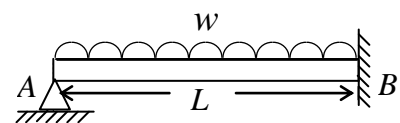
$$\text{Here } w_x = w \Rightarrow \therefore EI \cdot \frac{d^4 y}{dx^4} = w,$$

$$\text{or } \frac{d^4 y}{dx^4} = \frac{w}{EI}.$$

Integrating both sides directly gives,

$$\frac{d^3 y}{dx^3} = \frac{wx}{EI} + C_1 \Rightarrow \frac{d^2 y}{dx^2} = \frac{wx^2}{2EI} + C_1x + C_2,$$

$$\frac{dy}{dx} = \frac{wx^3}{6EI} + \frac{C_1x^2}{2} + C_2x + C_3 \Rightarrow y = \frac{wx^4}{24EI} + \frac{C_1x^3}{6} + \frac{C_2x^2}{2} + C_3x + C_4. \quad (\text{G.S})$$



(Statically indeterminate)

Boundary conditions,

$$1- y(0) = 0 \quad \Rightarrow \quad 0 = 0 + C_4 \quad \Rightarrow \quad C_4 = 0.$$

$$2- y''(0) = 0 \quad \Rightarrow \quad 0 = 0 + C_2 \quad \Rightarrow \quad C_2 = 0.$$

$$3- y(L) = 0 \quad \Rightarrow \quad 0 = \frac{wL^4}{24EI} + \frac{C_1 L^3}{6} + C_3 L. \quad \dots\dots\dots (1)$$

$$4- y'(L) = 0 \quad \Rightarrow \quad 0 = \frac{wL^3}{6EI} + \frac{C_1 L^2}{2} + C_3. \quad \dots\dots\dots (2)$$

Solving Eqs. (1) & (2) simultaneously yields,

$$C_1 = -\frac{3wL}{8EI} \quad \text{and} \quad C_3 = \frac{wL^3}{48EI}.$$

$$\therefore y = \frac{wx^4}{24EI} - \frac{wLx^3}{16EI} + \frac{wL^3x}{48EI}. \quad \text{(P.S)}$$

$$R_A = -EI(y''')_A = -EIy'''(0) = -EI\left(-\frac{3wL}{8EI}\right) = \frac{3wL}{8}. \quad (\uparrow)$$

$$R_B = -EI(y''')_B = -EIy'''(L) = -EI\left(\frac{wL}{EI} - \frac{3wL}{8EI}\right) = -\frac{5wL}{8}. \quad (\uparrow)$$

$$M_B = -EI(y'')_B = -EIy''(L) = -EI\left(\frac{wL^2}{2EI} - \frac{3wL^2}{8EI}\right) = -\frac{wL^2}{8}. \quad (\curvearrowright)$$

Note,

We can solve the above 4th order DE as a non-homogeneous linear DE, as follows

$$D^4 y = 0 \quad \Rightarrow \quad m^4 = 0 \quad \Rightarrow \quad m_{1,2,3,4} = 0,$$

$$\therefore y_c = C_1 + C_2 x + C_3 x^2 + C_4 x^3.$$

$$\text{Let } y_p = A_o(x^4) \quad \Rightarrow \quad y_p = A_o x^4,$$

$$y'_p = 4A_o x^3 \quad \Rightarrow \quad y''_p = 12A_o x^2 \quad \Rightarrow \quad y'''_p = 24A_o x \quad \Rightarrow \quad y^{iv}_p = 24A_o,$$

Substituting,

$$24A_o = \frac{w}{EI} \quad \Rightarrow \quad A_o = \frac{w}{24EI} \quad \Rightarrow \quad y_p = \frac{wx^4}{24EI}.$$

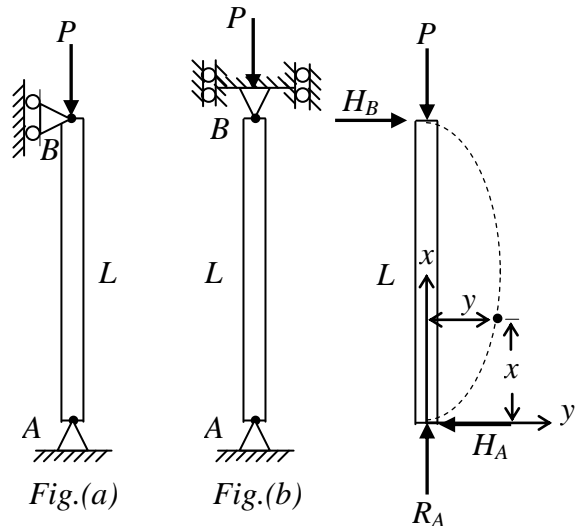
$$y = y_c + y_p \quad \Rightarrow \quad \therefore y = C_1 + C_2 x + C_3 x^2 + C_4 x^3 + \frac{wx^4}{24EI}. \quad \text{(G.S)}$$

2. Buckling of columns

Example 1: Determine the critical buckling load of a hinged-hinged column.

Solution:

Consider a column of length L , as shown in Fig.(a) or Fig.(b), hinged at both ends, and subjected to a compressive axial force P .



$$EI \cdot \frac{d^2 y}{dx^2} = -M_x. \quad \text{But } M_x = P \cdot y,$$

$$\therefore EI \cdot \frac{d^2 y}{dx^2} = -P \cdot y \Rightarrow \frac{d^2 y}{dx^2} + \frac{P}{EI} y = 0.$$

$$\text{Let } \beta^2 = \frac{P}{EI} \Rightarrow \frac{d^2 y}{dx^2} + \beta^2 y = 0,$$

$$\text{or } (D^2 + \beta^2)y = 0 \Rightarrow m^2 + \beta^2 = 0,$$

$$\Rightarrow m^2 = -\beta^2 \Rightarrow m_{1,2} = \pm \beta i,$$

$$\therefore y = C_1 \cos \beta x + C_2 \sin \beta x. \quad (\text{G.S})$$

Boundary conditions,

$$1. y(0) = 0 \Rightarrow 0 = C_1 + 0 \Rightarrow C_1 = 0.$$

$$\therefore y = C_2 \sin \beta x.$$

$$2. y(L) = 0 \Rightarrow 0 = C_2 \sin \beta L.$$

$$\text{If } C_2 = 0 \Rightarrow y = 0.$$

(i.e. the column remains straight)

$$\therefore C_2 \neq 0 \Rightarrow \sin \beta L = 0 \Rightarrow \beta L = 0, \pi, 2\pi, \dots, n\pi,$$

$$\therefore \beta L = n\pi \Rightarrow \beta = \frac{n\pi}{L}. \quad (n = 1, 2, 3, \dots)$$

$$\text{But } \beta^2 = \frac{P}{EI} \Rightarrow \frac{P}{EI} = \frac{n^2 \pi^2}{L^2} \Rightarrow P = \frac{n^2 \pi^2 EI}{L^2}.$$

$$\text{For } n = 1 \Rightarrow P_{cr} = \frac{\pi^2 EI}{L^2}, \quad (\text{Euler load or critical load})$$

$$\text{and } y = C_2 \sin \frac{\pi x}{L}. \quad (C_2 \text{ remains indeterminate, that is } y(\frac{L}{2}) = C_2)$$

To determine M_x :

Either from down (left);

$$M_x = R_A \cdot y + H_A \cdot x.$$

$$\sum F_x = 0,$$

$$R_A - P = 0 \Rightarrow R_A = P.$$

$$\sum F_y = 0,$$

$$H_B - H_A = 0 \Rightarrow H_A = H_B.$$

$$\sum (M)_B = 0,$$

$$H_A \cdot L = 0 \Rightarrow H_A = 0,$$

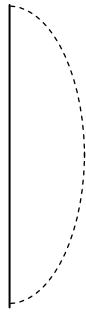
$$\therefore H_B = 0.$$

$$\therefore M_x = P \cdot y.$$

Or from up (right);

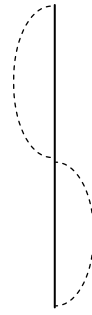
$$M_x = P \cdot y - H_B \cdot (L - x)$$

$$= P \cdot y.$$



$$n = 1$$

$$P_{cr} = \frac{\pi^2 EI}{L^2}$$



$$n = 2$$

$$P_{cr} = \frac{4\pi^2 EI}{L^2}$$



$$n = 3$$

$$P_{cr} = \frac{9\pi^2 EI}{L^2}$$

Critical buckling load: is the smallest value of axial load that can cause buckling.

* If $P < P_{cr}$, then the case is “*stable equilibrium*”. In this case, no buckling would occur. If lateral deflection is produced, by a horizontal force, then this deflection vanishes when the horizontal force is removed.

* If $P = P_{cr}$, then the case is “*neutral equilibrium*”. In this case, small and limited buckling may occur. If lateral deflection is produced, by a horizontal force, then this deflection remains constant even when the horizontal force is removed.

* If $P > P_{cr}$, then the case is “*unstable equilibrium*”. In this case, large not-controlled buckling may occur. If lateral deflection is produced, by a horizontal force, then this deflection will be increased, and if not controlled, the column will collapse.

Example 2: Determine the critical buckling load of a fixed-free column.

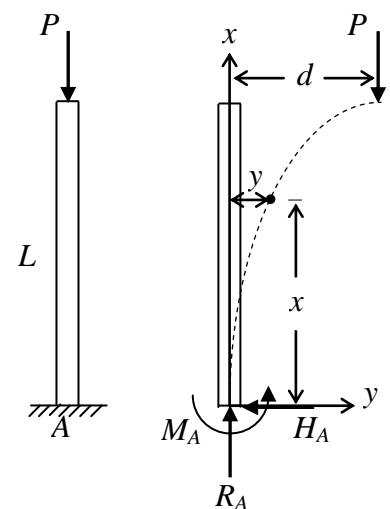
Solution:

Let the buckling at the free end is (d).

$$EI \cdot \frac{d^2 y}{dx^2} = -M_x. \quad \text{But } M_x = -P(d - y),$$

$$\therefore EI \cdot \frac{d^2 y}{dx^2} = -[-P(d - y)] \Rightarrow \frac{d^2 y}{dx^2} + \frac{P}{EI} y = \frac{P}{EI} d.$$

$$\text{Let } \beta^2 = \frac{P}{EI} \Rightarrow \frac{d^2 y}{dx^2} + \beta^2 y = \beta^2 d,$$



or $(D^2 + \beta^2)y = \beta^2 d \Rightarrow m^2 + \beta^2 = 0,$

$\Rightarrow m^2 = -\beta^2 \Rightarrow m_{1,2} = \pm \beta i,$

$\therefore y_c = C_1 \cos \beta x + C_2 \sin \beta x.$

Let $y_p = A \Rightarrow y'_p = y''_p = 0.$

Substituting,

$0 + \beta^2 A = \beta^2 d \Rightarrow A = d \Rightarrow y_p = d.$

$y = y_c + y_p,$

$\therefore y = C_1 \cos \beta x + C_2 \sin \beta x + d. \quad \text{(G.S)}$

Boundary conditions,

1. $y(0) = 0 \Rightarrow 0 = C_1 + d \Rightarrow C_1 = -d.$

$\therefore y = -d \cos \beta x + C_2 \sin \beta x + d.$

2. $y'(0) = 0, \quad y' = -\beta C_1 \sin \beta x + \beta C_2 \cos \beta x \Rightarrow 0 = 0 + \beta C_2 \Rightarrow C_2 = 0.$

$\therefore y = -d \cos \beta x + d \quad \text{or} \quad y = d(1 - \cos \beta x).$

3. $y(L) = d \Rightarrow d = d(1 - \cos \beta L) \Rightarrow 1 = 1 - \cos \beta L,$

$\Rightarrow \cos \beta L = 0 \Rightarrow \beta L = \frac{\pi}{2}, \frac{3\pi}{2}, \dots, \frac{(2n-1)\pi}{2},$

$\therefore \beta L = \frac{(2n-1)\pi}{2} \Rightarrow \beta = \frac{(2n-1)\pi}{2L}. \quad (n = 1, 2, 3, \dots)$

But $\beta^2 = \frac{P}{EI} \Rightarrow \frac{P}{EI} = \frac{(2n-1)^2 \pi^2}{4L^2} \Rightarrow P_{cr} = \frac{(2n-1)^2 \pi^2 EI}{4L^2}.$

For $n = 1 \Rightarrow P_{cr} = \frac{\pi^2 EI}{4L^2}. \quad \text{(i.e. } \frac{1}{4} \text{ times the critical load for hinged-hinged case)}$

To determine M_x :
 Either from up (right);
 $M_x = -P(d - y).$
 Or from down (left);
 $M_x = R_A \cdot y + H_A \cdot x - M_A.$
 $\sum F_x = 0,$
 $R_A - P = 0 \Rightarrow R_A = P.$
 $\sum F_y = 0,$
 $-H_A = 0 \Rightarrow H_A = 0.$
 $\sum (M)_A = 0,$
 $M_A - Pd = 0 \Rightarrow M_A = Pd.$
 $\therefore M_x = Py - Pd$
 $\qquad \qquad \qquad = -P(d - y)$